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MSI3200G

Inertial Measurement Unit

User Manual

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1 FEATURES

- Miniature package
- Low noise
- Low bias instability
- Excellent performance in vibration and shock environments
- 6 axes offered in same package
- Electronically calibrated axis alignment
- Gyros based on MT MicroSystem Single-crystal silicon technology
- No intrinsic wear-out effects
- Insensitive to magnetic fields
- Digital interface, RS422
- Continuous self-diagnostics

2 GENERAL DESCRIPTION

MSI3200G is a MEMS inertial measurement unit consisting of 3 high accuracy MEMS-based gyros, 3 high stability accelerometers, power switching circuit, software of IMU, internal and external mechanical structure of energy dissipation structures and shock absorber in a miniature package.

Both the gyros and the accelerometers are based on advanced MEMS Technology and wafer-level-package, which is highly stable and reliable. The MEMS chip, ASIC and temperature sensor are integrated and assembled in one ceramic package.

For gyroscopes, they are used to measure the palstance on three orthonormal axes of sensing carrier, accelerometers are used to compensation to enhance the sensitivity of accelerate value on three orthonormal axes. Digital circuit and the software are used to compensation to the error of component and transfer the message back to the computer control loop installed on the missile through the RS422 interface. Each axis is factory-calibrated for bias, sensitivity and compensated for temperature effects to provide high-accuracy measurements in the temperature range -45°C to $+85^{\circ}\text{C}$. The unit runs off a single $+12\text{V}$ supply. Meanwhile, MSI3200G could not only self-check the operative condition of inertial component and digital circuit before launch but also continuous diagnose the operative condition of the product during operation.

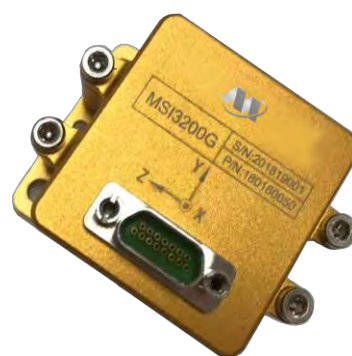


Fig.2-1 Image of MSI3200G

3 Typical Applications

- Platform stabilization
- Navigation and dead reckoning
- Dynamic attitude measurement
- Ship motion measurement

4 DEFINITIONS AND ABBREVIATIONS USED IN DOCUMENT

4.1 Definitions

$g_0 = 9.79973 \text{ m/s}^2$ (standard gravity)

4.2 Abbreviations

Table4-1:Abbreviations

ABBREVIATION	FULL NAME
FS	Full-Scale
LF	Line Feed
tbd	to be defined

5 ABSOLUTE MAXIMUM RATINGS

Stresses beyond those listed in Table 5-1 may cause permanent damage to the device.

Exposure to any Absolute Maximum Rating condition for extended periods may affect device reliability and lifetime.

Table 5-1: Absolute maximum ratings

Parameter	Rating	Comment
Mechanical shock	1 500g/1ms half-sine	Any direction. Ref: MIL STD-883G
Storage temperature	-55°C to +90°C	
+12V to GND	9V to 15V	
RX+ or RX- to GND	-0.3V to 3.6V	
RX+ to RX-	±3.3V	
TX+ or TX- to GND	-0.3V to 3.6V	

6 SPECIFICATIONS

6.1 SYSTEM CHARACTERISTICS

Table 6-1: Operating conditions

Parameter	Conditions	Min	Nom	Max	Unit	Note
INPUT RANGE, ANGULAR RATE		±300			%s	
INPUT RANGE, ACCELERATION		±15			g	
POWER SUPPLY		9	12	15	V	
OPERATING TEMPERATURE		-40		+85	°C	

Table 6-2: Functional specifications, general

Parameter	Conditions	Min	Nom	Max	Unit	Note
POWER CONSUMPTION						
Power consumption			1.2	1.5	W	
TIMING						
Start-Up time after Power-On				1	s	1
Time to valid data	T=+25°C			2	s	2
RS422 Bit-Rate		921600			bps	
Sample Rate			400		Samples/s	
RS422 PROTOCOL						
Start Bit			1		bit	
Data Length			8		bits	
Parity			EVEN			
Stop Bits			1		bit	
CHASSIS						
Isolation resistance chassis to						
GND(pin 15)	500V		20		MΩ	

Note 1: Time from Power-On to start of datagram transmissions (starting with part-number datagram)

Note 2: Time from Power-On to the reset of the Start-Up. During this period the output data should be regarded as non-valid.

Table 6-3: Functional specifications, gyros

Parameter	Conditions	Min	Nom	Max	Unit	Note
GYRO						
Full Scale(FS)		±300			°/s	
Resolution			32		bits	
			0.36		°/h	
Non-Linearity			200		ppm	
Bandwidth(-3dB)			220		Hz	
Sample Rate			400		Samples/s	
Group Delay			3		ms	
Bias Range		-180	0	180	°/h	
Bias stability			10		°/h	
Bias error over temperature gradients	$\Delta T < \pm 1^\circ\text{C}/\text{min}$		5		°/h	
Bias Instability	Allan Variance @25°C		3		°/h	
Angular Random Walk	Allan Variance @25°C		0.3		°/Vhr	
Linear Acceleration Effect	With g-compensation		0.005		°/s/g	
Misalignment			1		mrad	

Table 6-4: Functional specifications, accelerometers

Parameter	Conditions	Min	Nom	Max	Unit	Note
ACCELEROMETER						
Full Scale(FS)			±15		g	
Resolution			32		bits	
			16.7		ug	
Scale Factor 1 year stability			300	1000	ppm	
Non-Linearity			1000		ppm	
Bandwidth(-3dB)		100	147		Hz	
Sample Rate			400		Samples/s	
Group Delay				5	ms	
Bias switch on/off repeatability		-0.5		0.5	mg	
Bias 1 year stability				2	mg	
Bias Instability	Allan Variance @25°C			0.1	mg	
Velocity Random Walk	Allan Variance @25°C			0.05	m/s/vh	
Misalignment			1		mrad	

Note1: For lager scales of accelerator, you should consult producers for more information.

6.2 TYPICAL PERFORMANCE CHARACTERISTICS

6.2.1 GYRO CHARACTERISTICS

6.2.1.1 Root Allan Variance of gyro

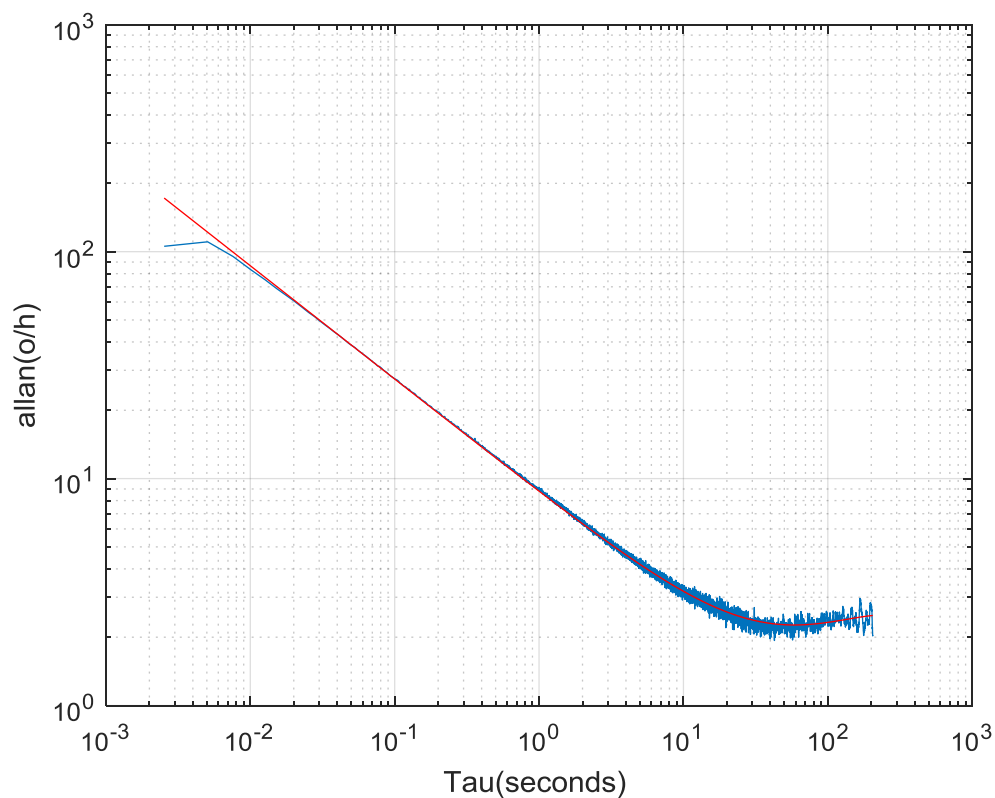


Figure 6-1: Typical Allan-Variance of gyro-X 250°/s gyro)

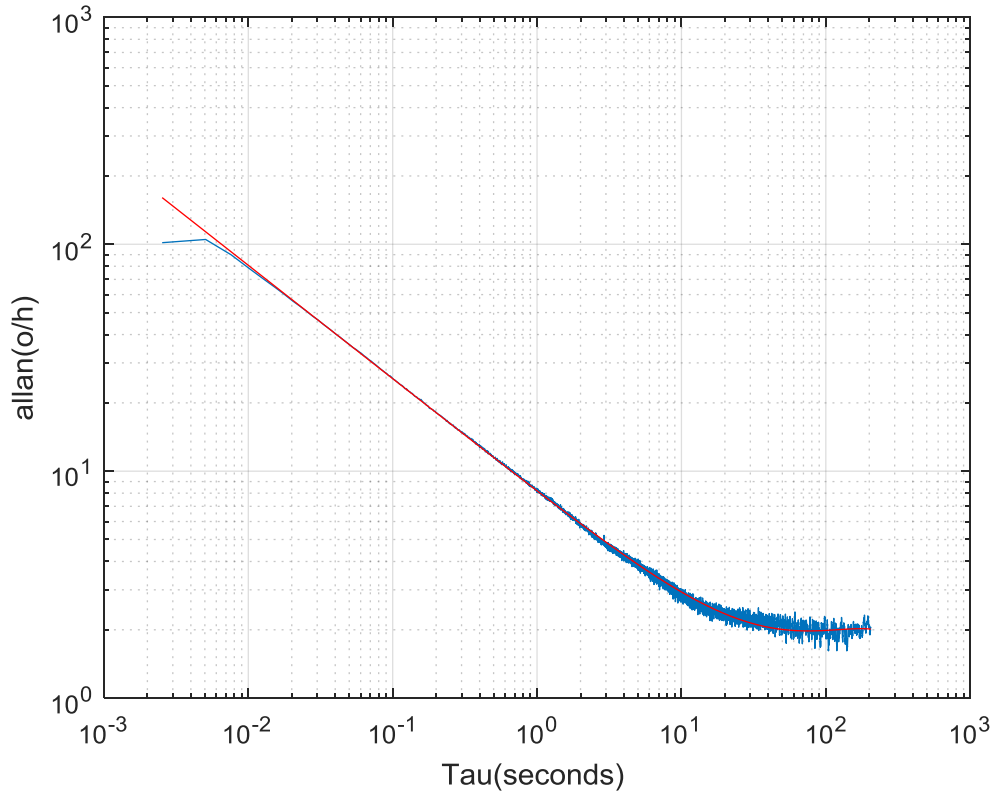


Figure 6-2: Typical Allan-Variance of gyro-Y (250°/s gyro)

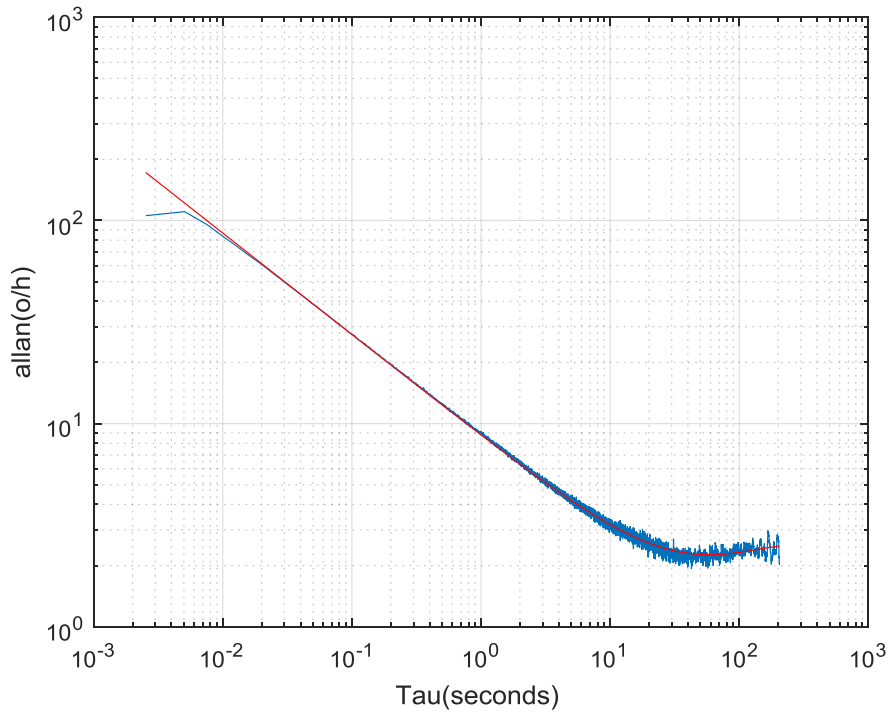


Figure 6-3: Typical Allan-Variance of gyro-Z (250°/s gyro)

6.2.1.2 Initial bias drift of gyro

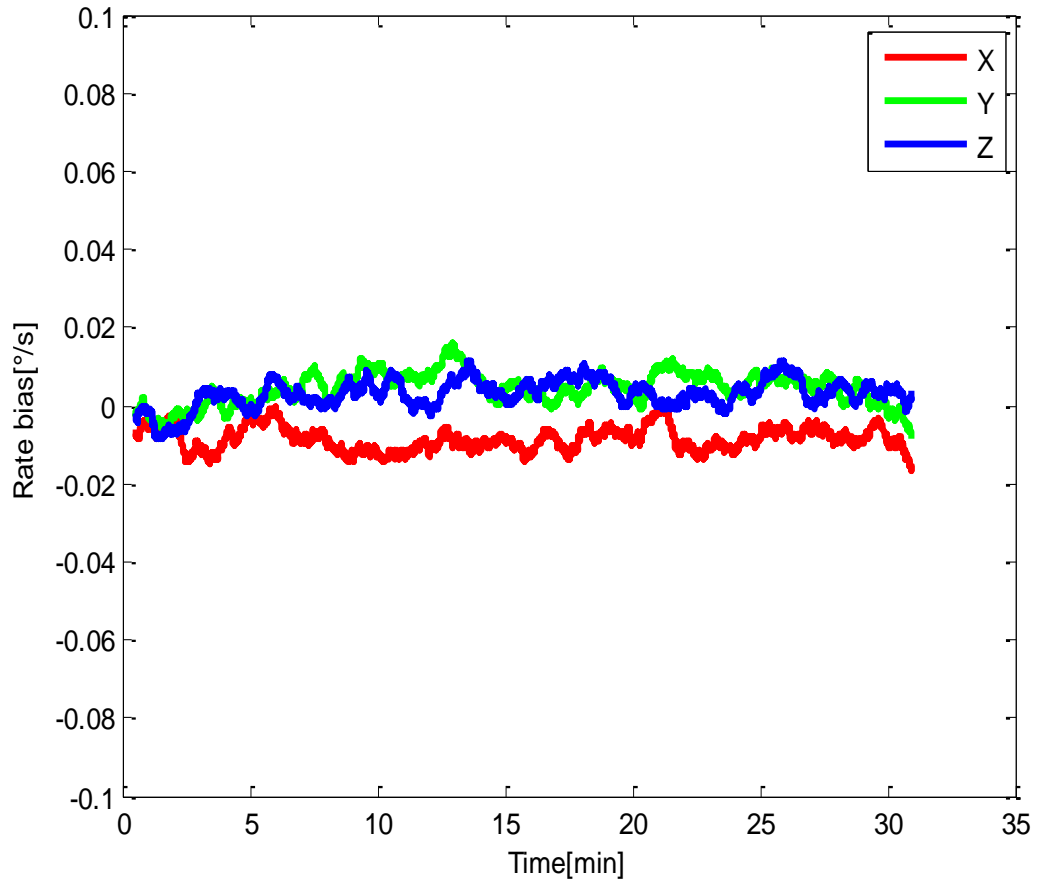


Figure 6-4: Typical normalized initial bias drift of gyro (250°/s gyro)

6.2.1.3 Bias drift over temperature

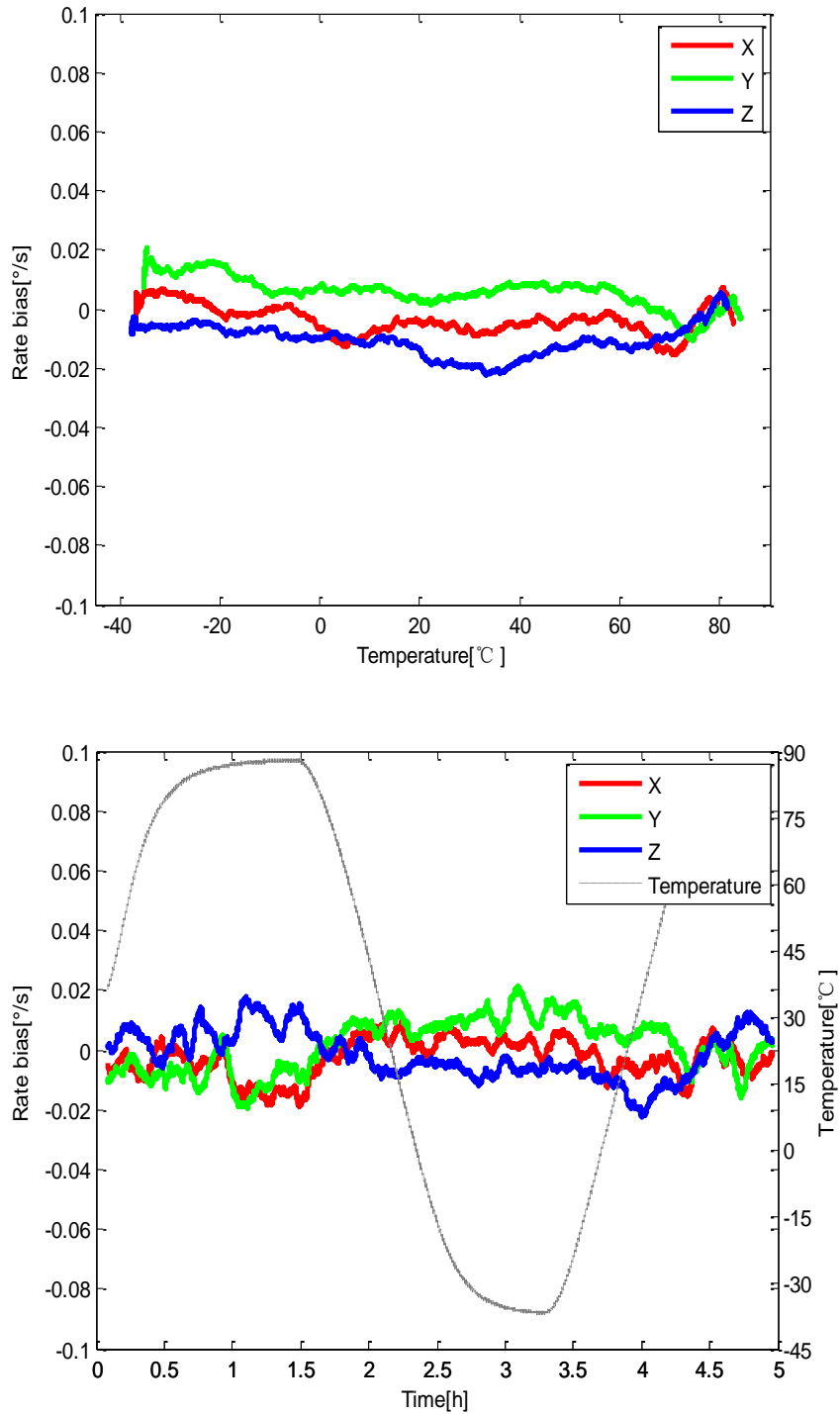


Figure 6-5: Typical Bias drift over temperature (250°/s gyro)

6.2.2 ACCELEROMETER CHARACTERISTICS

6.2.2.1 Initial bias drift of accelerometer

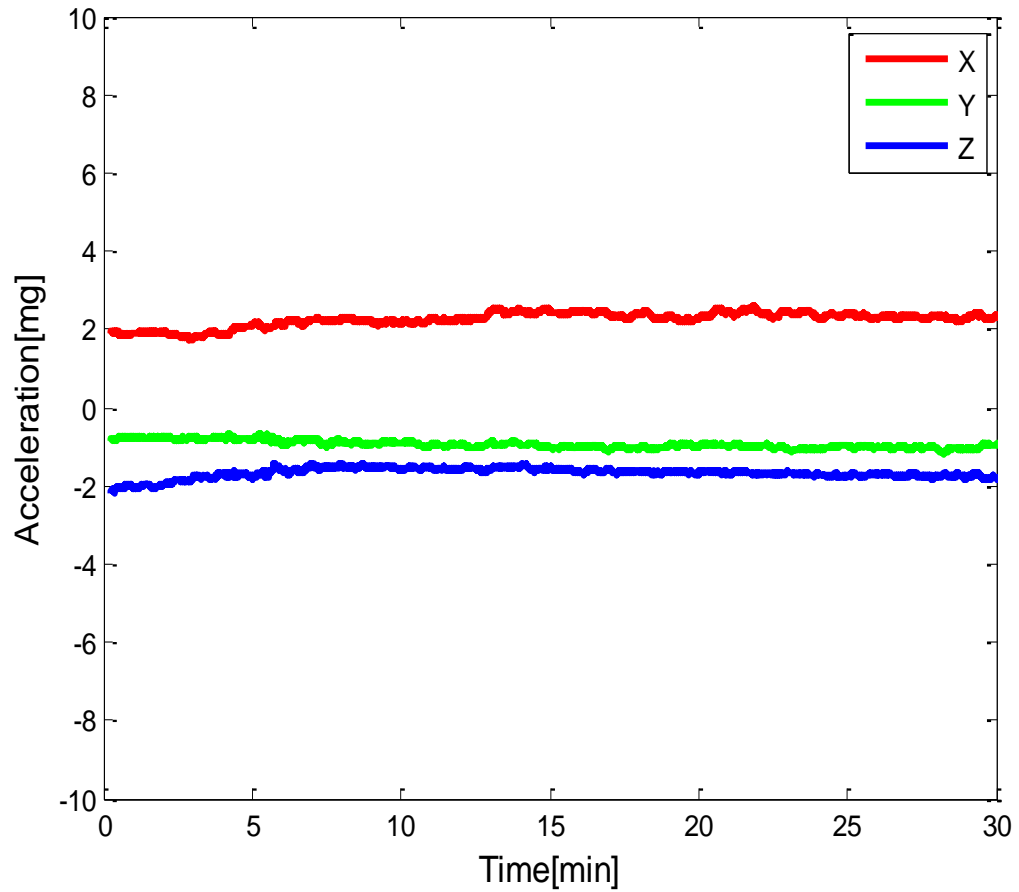


Figure 6-6: Typical normalized initial bias drift of accelerometer (15g accelerometer)

6.2.2.2 Bias drift over temperature

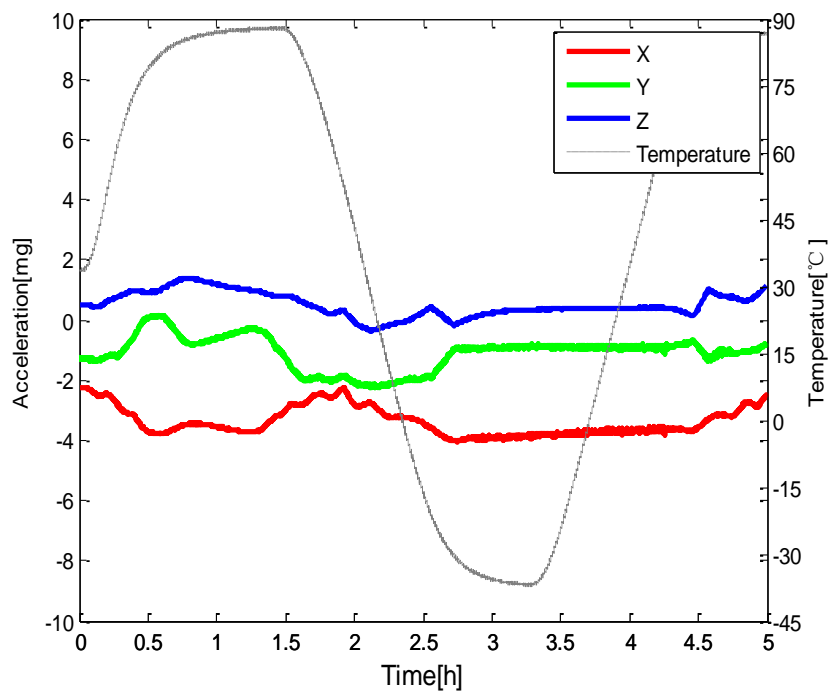
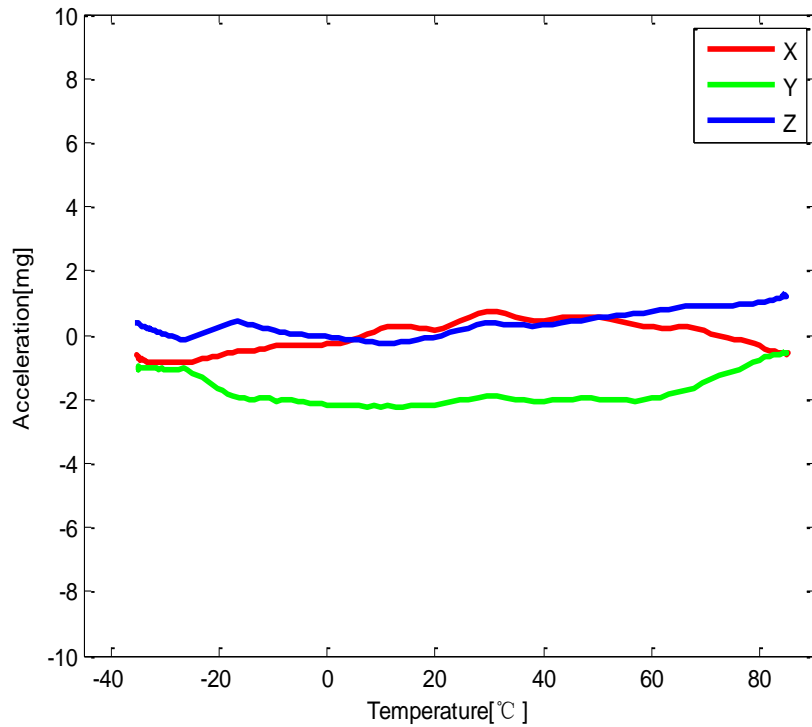


Figure 6-7: Typical Bias drift over temperature (15g accelerometer)

6.3 Datagram specifications

6.3.1 Normal Mode datagram

Table 6-5: Specification of the Normal Mode

Byte#	Bit#								Specification
	7	6	5	4	3	2	1	0	
0	1	0	1	0	0	1	0	1	Normal Mode datagram identifier for Normal Mode datagram with full content(0xA5).
1	0	0	0	1	1	1	0	1	The length of the datagram(0x1D).
2	n7	n6	n5	n4	n3	n2	n1	n0	Counter, ref. section 8.1.4
3	n 15	n 14	n 13	n 12	n 11	n 10	n 9	n 8	
4	Gx7	Gx6	Gx5	Gx4	Gx3	Gx2	Gx1	Gx0	X-axis gyro output, ref. section 8.1.1 for conversion to units
5	Gx15	Gx14	Gx13	Gx12	G11	Gx10	Gx9	Gx8	
6	Gx23	Gx22	Gx21	Gx20	Gx19	Gx18	Gx17	Gx16	
7	Gx31	Gx30	Gx29	Gx28	Gx27	Gx26	Gx25	Gx24	
8	Gy7	Gy6	Gy5	Gy4	Gy3	Gy2	Gy1	Gy0	Y-axis gyro output, ref. section 8.1.1 for conversion to units
9	Gy15	Gy14	Gy13	Gy12	G11	Gy10	Gy9	Gy8	
10	Gy23	Gy22	Gy21	Gy20	Gy19	Gy18	Gy17	Gy16	
11	Gy31	Gy30	Gy29	Gy28	Gy27	Gy26	Gy25	Gy24	
12	Gz7	Gz6	Gz5	Gz4	Gz3	Gz2	Gz1	Gz0	Z-axis gyro output, ref. section 8.1.1 for conversion to units
13	Gz15	Gz14	Gz13	Gz12	G11	Gz10	Gz9	Gz8	
14	Gz23	Gz22	Gz21	Gz20	Gz19	Gz18	Gz17	Gz16	
15	Gz31	Gz30	Gz29	Gz28	Gz27	Gz26	Gz25	Gz24	
16	Ax7	Ax6	Ax5	Ax4	Ax3	Ax2	Ax1	Ax0	X-axis accelerometer output, ref. section 8.1.2 for conversion to units
17	Ax15	Ax14	Ax13	Ax12	Ax11	Ax10	Ax9	Ax8	
18	Ax23	Ax22	Ax21	Ax20	Ax19	Ax18	Ax17	Ax16	
19	Ax31	Ax30	Ax29	Ax28	Ax27	Ax26	Ax25	Ax24	
20	Ay7	Ay6	Ay5	Ay4	Ay3	Ay2	Ay1	Ay0	Y-axis accelerometer output, ref. section 8.1.2 for conversion to units
21	Ay15	Ay14	Ay13	Ay12	G11	Ay10	Ay9	Ay8	
22	Ay23	Ay22	Ay21	Ay20	Ay19	Ay18	Ay17	Ay16	
23	Ay31	Ay30	Ay29	Ay28	Ay27	Ay26	Ay25	Ay24	
24	Az7	Az6	Az5	Az4	Az3	Az2	Az1	Az0	Z-axis accelerometer output, ref. section 8.1.2 for conversion to units
25	Az15	Az14	Az13	Az12	Az11	Az10	Az9	Az8	
26	Az23	Az22	Az21	Az20	Az19	Az18	Az17	Az16	
27	Az31	Az30	Az29	Az28	Az27	Az26	Az25	Az24	
28	T7	T6	T5	T4	T3	T2	T1	T0	temperature data, ref. section 8.1.3for conversion to units
29	T15	T14	T13	T12	T11	T10	T9	T8	
30	c7	c6	c5	c4	c3	c2	c1	c0	Sum Check is performed on all preceding bytes, ref. section 6.3.2
31	c15	c14	c13	c12	c11	c10	c9	c8	



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6.3.2 Sum Check

At the end of all datagrams is a 16-bit Checksum. The Sum checksum enables the user to detect errors in the transfer of data from MSI3200G. The Sum is calculated using the following equation: seed = 0x0000; Sum from the first byte to the twenty-ninth byte.

7 MECHANICAL

Table 7-1: Mechanical specifications

Parameter	Conditions	Min	Nom	Max	Unit	NOTE
HOUSING MATERIAL		Aluminium Alloy 6082-T6,DIN EN 754-2				
SURFACE TREATMENT Passivation		Surtec 650				
WEIGHT		100	106.5	110	grams	
VOLUME		67.3 4.11			ccm cu in	
CONNECTOR Type Number of pins Contact type female		J30J-15ZKP 15 female				
PLUG Proposed plug to fit connector		Axon MDA 2 15				
Proposed cover to fit plug	For best EMI performance	Axon micro-D EMI back shell				
FIXATION BOLTS Recommended torque Steel base		ISO 4762 / DIN 912 3.5			Nm	
	Aluminium base	3.0			Nm	

7.1 Mechanical dimensions

Mechanical Drawing (All dimensions are in mm.)

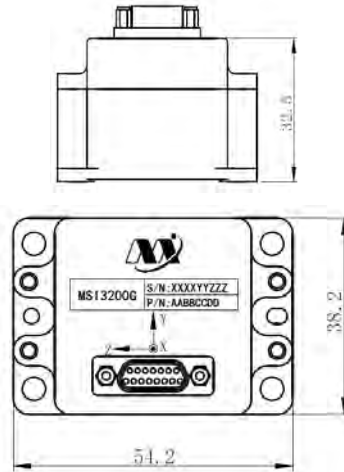


Figure 7-1: Mechanical dimensions

7.2 Pin configuration

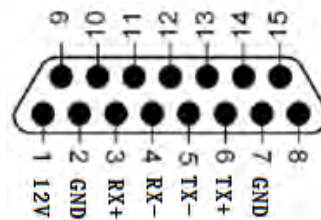


Figure 7-2: Pin configuration as seen from front of MSI3200G

The output connector is J30J-15ZKP, the Electric connection definition is as follows:

Table 7-2: Pin descriptions

ID	Definition	Note
1	+12V	Power
2	GND	GND
3	RX+	RS422 Receive Positive
4	RX-	RS422 Receive Negative
5	TX-	RS422 Transmit Negative
6	TX+	RS422 Transmit Positive
7	RS422 GND	RS422 GND
8~15	Blank	N/A

7.3 Definition of axes

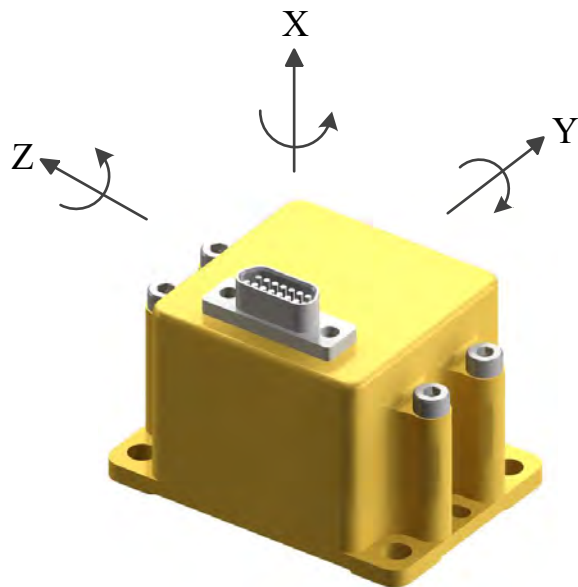


Figure 7-3: Definition of axes

8 BASIC OPERATION

MSI3200G is very simple to use. The unit will start performing measurements and transmit the results over the RS422 interface without any need for additional signaling or set-up after power-on. Figure 8-1 shows the simplest connection set-up for MSI3200G.

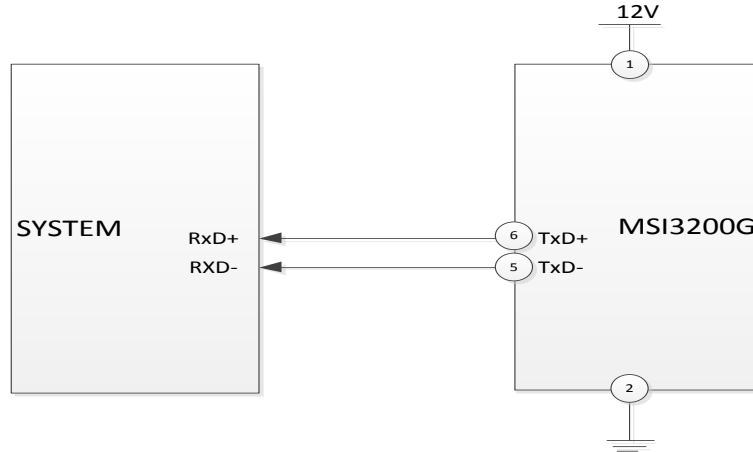


Figure 8-1: Full Function Electrical Connection Diagram

MSI3200G will constantly measure the available sensor channels at the configured sample rate. MSI3200G will continue to transmit data.

8.1 Data output options and interpretation

8.1.1 Gyro output unit = Angular Rate

In the case of MSI3200G being configured to output angular rate, Following equation and figure show how to convert to [°/s]. Please note that the output data is represented as two's complement.

Equation 1: Converting output to [°/s]:

$$Output[°/s] = \frac{AR_1 \cdot 2^{24} + AR_2 \cdot 2^{16} + AR_3 \cdot 2^8 + AR_4}{10000}$$

where AR₁ is the most significant byte of the 32bit output
 AR₂ is the second byte of the 32bit output
 AR₃ is the third byte of the 32bit output
 AR₄ is the least significant byte of the 32bit output



Figure 8-2: Converting output bytes to [°/s]

8.1.2 Accelerometer output unit = Acceleration

Equation 2: Converting output to [g]:

$$Output[g] = \frac{AR_1 \cdot 2^{24} + AR_2 \cdot 2^{16} + AR_3 \cdot 2^8 + AR_4}{60000}$$

where AR₁ is the most significant byte of the 32bit output
 AR₂ is the second byte of the 32bit output
 AR₃ is the third byte of the 32bit output
 AR₄ is the least significant byte of the 32bit output



Figure 8-3: Converting output bytes to [g]

8.1.3 Temperature

Temperature data is available in certain datagrams (ref. section 12).

Following equation and figure show how to convert to [°C]. Please note that the output data is represented as two's complement.

Equation 3: Converting temperature data to [°C]

$$Output[°C] = \frac{T_1 \cdot 2^8 + T_2 - 6527}{79.68}$$

where T₁ is the most significant byte of the 16bit output
 T₂ is the least significant byte of the 16bit output



Figure 8-4: Converting temperature data to [°C]

8.1.4 Counter

Counter is continuously counting the internal samples (400 samples/s). Counter is an un-signed word taking values in the interval [0, 65535]. The counter will naturally wrap-around.

9 MARKING

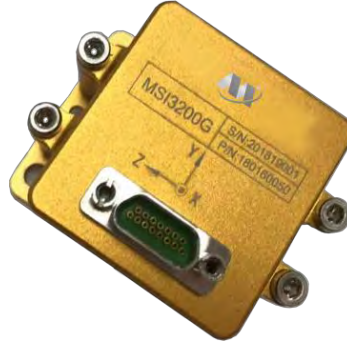


Figure 9-1: Example of marking of MSI3200G